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Concept Design and Performance Evaluation of UAV-based Backhaul Link with Antenna Steering

Jiri Pokorny, Aleksandr Ometov, Pablo Pascual, Carlos Baquero, Pavel Masek, Alexander Pyattaev, Ainoa Garcia, Carlos Castillo, Sergey Andreev, Jiri Hosek, and Yevgeni Koucheryavy

Abstract: At present, cellular coverage in many rural areas remains intermittent. Mobile operators may not be willing to deploy expensive network infrastructure to support low-demand regions. For that reason, solutions for the rapid deployment of base stations in areas with insufficient or damaged operator infrastructure are emerging. Utilization of unmanned aerial vehicles (UAVs) or drones serving as data relays holds significant promise for delivering on-demand connectivity as well as providing public safety services or aiding in recovery after communication infrastructure failures caused by natural disasters. The use of UAVs in provisioning high-rate radio connectivity and bringing it to remote locations is also envisioned as a potential application for fifth-generation (5G) communication systems. In this study, we introduce a prototype solution for an aerial base station, where connectivity between a drone and a base station is provided via a directional microwave link. Our prototype is equipped with a steering mechanism driven by a dedicated algorithm to support such connectivity. Our experimental results demonstrate early-stage connectivity and signal strength measurements that were gathered with our prototype. Our results are also compared against the free-space model. These findings support the emerging vision of aerial base stations as part of the 5G ecosystem and beyond.

Index Terms: Drone, antenna steering, prototype, measurements.

I. Introduction

MOBILE traffic demand has been dramatically increasing over the past decades [1] and is predicted to continue growing with the evolution of the next-generation (5G) wireless networks [2, 3]. Even though demand for cellular coverage in rural areas is high, access therein is still limited for many people [4]. According to BusinessLIVE, this is primarily because cellular operators focus on developing/upgrading their installations in metropolitan areas, where demand for spectrum is higher [5].

From another perspective, a natural disaster may occur and

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J. Pokorny, P. Masek, and J. Hosek are with Department of Telecommunications, Brno University of Technology, Brno, Czech Republic.

A. Ometov, P. Pascual, C. Baquero, S. Andreev, and Y. Koucheryavy are with Laboratory of Electronics and Communications Engineering, Tampere University of Technology, Tampere, Finland.

A. Pyattaev, A. Garcia Marti, and C. Castillo Mateos are with YL-Verkot Oy, Tampere, Finland.

A. Ometov is the corresponding author: aleksandr.ometov@tut.fi

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In urban areas, where coverage is already in place, the demand for cellular connectivity is also growing. This is caused by the increased use of mobile devices and the demand for faster data rates. However, in rural areas, the demand for cellular coverage is even more pressing. According to BusinessLIVE, this is primarily because cellular operators focus on developing/upgrading their installations in metropolitan areas, where demand for spectrum is higher [5].

For that reason, solutions for the rapid deployment of base stations in areas with insufficient or damaged operator infrastructure are emerging. Utilization of unmanned aerial vehicles (UAVs) or drones serving as data relays holds significant promise for delivering on-demand connectivity as well as providing public safety services or aiding in recovery after communication infrastructure failures caused by natural disasters. The use of UAVs in provisioning high-rate radio connectivity and bringing it to remote locations is also envisioned as a potential application for fifth-generation (5G) communication systems. In this study, we introduce a prototype solution for an aerial base station, where connectivity between a drone and a base station is provided via a directional microwave link. Our prototype is equipped with a steering mechanism driven by a dedicated algorithm to support such connectivity. Our experimental results demonstrate early-stage connectivity and signal strength measurements that were gathered with our prototype. Our results are also compared against the free-space model. These findings support the emerging vision of aerial base stations as part of the 5G ecosystem and beyond.

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I. Introduction

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A. Related Work

The motivation for our research is strongly supported by the research community. The authors in [29] studied the coverage provided by low-altitude platforms for use in urban and rural disaster-affected areas. In [30], the authors proposed a statistical propagation model for the air-to-ground path loss between an LAP and a terrestrial terminal. Solutions with balloons are typically not very fast and flexible due to their slower deployment, lower mobility, and larger dimensions. Some prior research investigated opportunities for drone deployment in different scenarios. The authors of [31] focused on analyzing urban utilization of skyscraper drone BSs. This study modeled such a system, which is relatively close to the results collected in our study.

Overall, aerial BSs using UAVs with propellers, specifically, quadcopters, hexacopters, or octocopters, were also considered by other researchers. The recent work in [32] describes flying cell towers using UAVs with propellers with a GreenCell LTE BS that extends an existing cellular network or creates a new one. Furthermore, CyPhy company develops a tethered drone powered from the ground, which allows extension of currently limited flight times, but at the cost of lower mobility and more complex deployment. The proposed drones were developed for various purposes, one of them being communications [33]. The industrial giant Nokia has also developed a compact LTE solution that employs drones for public safety [34].

The work in [35] presents an innovative approach for enabling emergency communications in the field based on a combination of aerial and terrestrial mobile stations. The project aims to considerably improve disaster recovery and crisis management. A similar concept is described in [36]. In parallel with prototyping, many researchers are working with aerial access points from a simulation perspective. The conventional aspects of capacity, coverage, and interference are discussed in [37, 38].

The overall benefits brought by utilizing aerial access points for cellular network enhancement are discussed in [39]. The work in [40] sheds some light on interference and propagation impact of cellular BSs. The authors of [41] developed an analytical model to evaluate drone-to-ground communications. In general, many telecommunication developers and researchers have the same goal: to bring network coverage to remote areas by employing an aerial vehicle. A drone can reach the designated area promptly while delivering connectivity to locations not accessible by conventional ground vehicles.

B. Main Contributions

In this study, we consider two types of scenarios shown in Fig. 1: (i) a public event (or on-demand connectivity) and (ii) a Public Protection and Disaster Relief (PPDR) situation [7, 42]. The first type of layout is usually known in advance and can be planned efficiently. Here, the advantage of the proposed solution is the cost of deployment. The second scenario can correspond to a natural disaster, which is unpredictable and requires an immediate response.

In this paper, we discuss our concept of an aerial BS, where backhaul (BH) link connectivity is implemented via microwave communication between the BS and the drone. The main contributions of our paper are as follows:

1. Concept design. We propose a new design for an aerial vehicle controlled via the TCP/IP protocol stack (IP layer). The introduced solution is capable of carrying network equipment for a BH link. This UAV was specially designed, constructed, and tested.

2. Easy-to-deploy LTE BS solution. The UAV-based eNodeB is flexible and can be deployed quickly. LTE relays are usually transported on trucks, which are slow and can potentially be stopped by damaged terrain and/or poor road infrastructure. The proposed solution can completely replace current alternatives due to its lower cost demands and significantly reduced reaction times in case of an unexpected situation, e.g., PPDR.

3. Steering algorithm and mechanism. The drone is a highly mobile aerial vehicle that cannot be fixed to any solid object. Its movement, whether intentional or not, can be random and not very precise. For this reason, we developed a steering algorithm that compensates for drone movement and maintains radio signal stability at high levels.

4. Early-stage measurements. We conducted initial connectivity and signal strength measurements with the utilized equipment and compared our results with the free-space propagation model.
II. Proposed System Architecture

Before continuing with the actual steering mechanism description, it is important to construct an initial setup for the UAV and the ground station. This initial composition will further facilitate the configuration of actual devices.

In this section, we describe the complete architecture of our developed testbed. It includes a description of the working UAV prototype together with the networking equipment and ground station. The overall setup of our proposed system is shown in Fig. 2. It includes the BH link, which handles two data streams: (i) one stream transmits user data and information (red link) and (ii) another carries telemetry information and provides drone control (blue link). Those links are established via a routing device located on the UAV side. The aerial vehicle is equipped with 5 GHz antennas to establish a BH link between the UAV and the ground station. The utilization of higher frequencies may also be considered [43].

Fig. 2. Configuration of the backhaul link and sub-links allocated within (control link and service link).

A crucial feature of the utilized UAV in our proposed test scenario is represented by a case where the UAV must move to a desired position and remain static by maintaining a connection to the ground station through a BH link. Therefore, the ground station can exchange data packets with users and telemetry packets with the UAV, as well as control the UAV in real time. To enhance the functionality of this link, a steering system is designed that autonomously tracks both antennas (detailed in Section III) in the UAV and the ground station. This system allows the UAV to be controlled over longer distances.

The following subsection discusses the hardware selected for the aerial vehicle and the ground station, along with a clarification of the relevant components.

A. UAV Prototype

Our UAV prototype has been designed similar to one completed by the authors in a prior robotic system project [44]. The key difference is that our current idea refers to an aerial vehicle instead of a terrestrial vehicle, thus resulting in different hardware requirements.

A diagram with the relevant hardware integrated into the UAV is detailed in Fig. 3(a). The primary equipment used in the UAV prototype is listed in Table 1.

B. Ground Station Configuration

The ground station includes a computer and an antenna providing the BH link. In our testbed, the computer is connected to the UAV via the BH link to control the drone and deliver connectivity. Fig. 3(b) shows the hardware used in the ground station. Since the tests in this project were carried out in the field, an extra battery was added to power the hardware on the ground station side. The primary hardware used in our ground station is listed in Table 2.

In what follows, we describe our autonomous steering system that enables bidirectional operation of the BH link.

III. Automatic Antenna Steering System

A. General Description

One of the solutions to extend coverage with the BH link and maintain signal quality at an appropriate level is the use of high gain and directional antennas at both sides of the link. The radiation pattern of such antennas is very narrow, and even a small misalignment may result in significant signal-to-noise ratio reduction. This gives rise to the requirement that both antennas are continuously pointed at each other. Hence, steering must be
The general scheme of our scenario and defines this problem. The error factor is negligible at such short distances. Fig. 4 shows the complexity but introduces a potential error factor. However, this error factor is negligible at such short distances. To achieve this, four servo motors were installed to provide mechanical rotation according to these calculated values. As we are considering short-to-medium distances for our prototype, we proceed with a flat-Earth approximation. This decreases the computational complexity but introduces a potential error factor. However, this error factor is negligible at such short distances. Fig. 4 shows the general scheme of our scenario and defines this problem.

To tackle the steering system, it is important to consider the parameters required to compute the elevation and azimuth angles, as explained in [45]. The final angles to steer the antennas can be obtained from the longitude, latitude, altitude, and heading of the UAV in the Earth-fixed coordinates. As we are adding another antenna to the UAV, one must consider the corresponding parameters for both antennas. The important parameters in our system are listed in Table 3.

The calculations required to obtain the final angles for the elevation and azimuth are performed as follows. The elevation angles for the ground station and the UAV are defined as

\[
\tan(\xi_1) = \frac{l - \sum_{i=1}^{n} h_i}{GCD} \quad \text{and} \quad \xi_1 = \arctan\left(\frac{l - \sum_{i=1}^{n} h_i}{GCD}\right). \quad (1)
\]

Here, GCD refers to the great-circle distance between the coordinates of the UAV and the GSA. This value can be computed using the Haversine expression:

\[
d = R_c, \quad (2)
\]

where \( R_c \) refers to the radius of the Earth, while \( c \) and \( a \) are computed as follows:

\[
c = 2\arctan2\left(\sqrt{a}, \sqrt{1-a}\right). \quad (3)
\]

where

\[
a = \sin^2\left(\frac{\phi_2 - \phi_1}{2}\right) + \cos(\phi_1)\cos(\phi_2)\sin^2\left(\frac{\lambda_2 - \lambda_1}{2}\right). \quad (4)
\]

Table 1. Main components of the UAV (refer to Fig. 3(a)).

<table>
<thead>
<tr>
<th>Element</th>
<th>Description</th>
</tr>
</thead>
<tbody>
<tr>
<td>1 Li-Po battery</td>
<td>Power supply for the UAV. Provides 14.8 V and has a capacity of 30,000 mAh.</td>
</tr>
<tr>
<td>2 Power module</td>
<td>DC-DC voltage regulator from 14.8 V to 5 V with maximum provided current of 3 A.</td>
</tr>
<tr>
<td>3 Flight controller</td>
<td>Module managing the rotation of each motor to control the trajectory of the UAV based on its position, axis state (roll, yaw, pitch), and commands issued by the operator from the ground station.</td>
</tr>
<tr>
<td>4 GPS and compass</td>
<td>Module providing the position and heading of the UAV.</td>
</tr>
<tr>
<td>5 Telemetry receiver and backup RC</td>
<td>Radio controller acting as a backup in case the BH link fails.</td>
</tr>
<tr>
<td>6 Beagle Bone</td>
<td>Microcomputer that provides a bridge between the IP layer and the flight controller.</td>
</tr>
<tr>
<td>7 On-board router</td>
<td>Device routing data and telemetry packets to the BH link.</td>
</tr>
<tr>
<td>8 On-board antenna</td>
<td>Antenna mounted on the UAV providing the BH link to the ground station. The antenna gain is 16 dBi and it communicates over the MikroTik Nv2 protocol at 5.5 GHz.</td>
</tr>
<tr>
<td>9 Servo motors</td>
<td>Electronic devices providing steering to the on-board antenna. The supported torque is 5.1 kg/cm.</td>
</tr>
</tbody>
</table>

Table 2. Main components of the ground station (refer to Fig. 3(b)).

<table>
<thead>
<tr>
<th>Element</th>
<th>Description</th>
</tr>
</thead>
<tbody>
<tr>
<td>1 Terminal</td>
<td>Computer controlling the UAV and handling the network in real time.</td>
</tr>
<tr>
<td>2 Ground station antenna</td>
<td>Antenna communicating with the on-board antenna of the UAV to provide the BH link. The device gain is 24.5 dBi. The antenna communicates over the MikroTik Nv2 protocol in the 5.5 GHz band.</td>
</tr>
<tr>
<td>3 Raspberry Pi</td>
<td>Microcomputer that provides a bridge between the IP layer and the flight controller.</td>
</tr>
<tr>
<td>4 Servo motors</td>
<td>Device routing data and telemetry packets to the BH link.</td>
</tr>
<tr>
<td>5 On-board antenna</td>
<td>Antenna mounted on the UAV providing the BH link to the ground station. The antenna gain is 16 dBi and it communicates over the MikroTik Nv2 protocol at 5.5 GHz.</td>
</tr>
<tr>
<td>6 Servo motors</td>
<td>Electronic devices providing steering to the on-board antenna. The supported torque is 5.1 kg/cm.</td>
</tr>
</tbody>
</table>

Table 3. Main parameters of the antenna steering system.

<table>
<thead>
<tr>
<th>Input parameters</th>
<th>Description</th>
</tr>
</thead>
<tbody>
<tr>
<td>( \xi_1 )</td>
<td>Elevation angle of the GSA</td>
</tr>
<tr>
<td>( \lambda_1 )</td>
<td>Longitude of the UAV</td>
</tr>
<tr>
<td>( \lambda_2 )</td>
<td>Longitude of the GSA</td>
</tr>
<tr>
<td>( l )</td>
<td>Relative altitude of the UAV</td>
</tr>
<tr>
<td>( h_i )</td>
<td>Altitude of the GSA</td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>Output parameters</th>
<th>Description</th>
</tr>
</thead>
<tbody>
<tr>
<td>( \xi_2 )</td>
<td>Elevation angle of the UAV antenna</td>
</tr>
<tr>
<td>( \theta_1 )</td>
<td>Bearing angle of the GSA</td>
</tr>
<tr>
<td>( \theta_2 )</td>
<td>Bearing angle of the UAV antenna</td>
</tr>
</tbody>
</table>

The notation and description for each parameter are as follows:

- \( \alpha \) and \( \beta \) are angles for the elevation and azimuth, respectively.
- \( \lambda_1 \) and \( \lambda_2 \) are the longitudes of the ground station antenna (GSA) and the UA V, respectively.
- \( \phi_1 \) and \( \phi_2 \) are the latitudes of the GSA and the UA V, respectively.
- \( h_i \) refers to the altitude of the ground station antenna (GSA) at each measurement.
- \( d \) is the great-circle distance between the GSA and the UA V.
- \( c \) is the maximum provided current of the servo motors.

Fig. 4. Computation of elevation and azimuth for the UAV antenna and the ground station antenna (GSA).
At this point, the elevation angle of the UAV’s on-board antenna is obtained directly from the following:

$$\xi_2 = 180 - 90 - \xi_1.$$  \tag{5}$$

The bearing angle $\theta$ of two GPS coordinate points expressed in latitude [radians] and longitude [radians] can be computed from

$$\theta = \arctan\left(\frac{\sin(\lambda_2 - \lambda_1)\cos(\phi_2)\cos(\phi_1)\sin(\phi_2)}{\cos(\phi_1)\cos(\phi_2)\cos(\lambda_2 - \lambda_1)}\right).$$  \tag{6}$$

This procedure is required for obtaining the bearing angle for both antennas. The antennas must be rotated by the angle computed above with respect to the North cardinal point. With this procedure, the four angles required to point the antennas are obtained. Note that the input parameters stated above must be known continuously at both sides of the link to compute the final elevation and bearing angles. As a result, communication must be established between the UAV and the GSA during initial synchronization. This transmission is allocated over the telemetry link.

The process described above is entirely autonomous and is executed in a loop, as explained in [46]. The implemented algorithm is executed repeatedly to continuously steer both antennas. A flowchart of the algorithm used to point both antennas is shown in Fig. 5.

B. Potential Sources of Errors

Once the autonomous steering system has been implemented, potential sources of errors that might affect the accuracy of the steering algorithm can be analyzed. This study may be considered as a base for future improvements in the real-world system.

It is important to remember that the implemented prototype is to be operated over short-to-medium distances (of below 1 km). Hence, only errors that become significant over long-distance communication might affect the accuracy of the steering system. The relevant sources of errors are sorted according to their expected impact:

- Geodesic calculations – Should the system be operated over longer distances, the calculations performed above are to be performed according to a geodesic model that takes into account the curvature of the Earth.
- GPS and compass inaccuracies – As explained in [47], GPS systems are susceptible to failure or may give inaccurate results due to surrounding obstacles, such as mountains, vegetation, or human-made structures.
- Altitude inaccuracies – The UAV is responsible for managing information regarding its altitude using a barometric sensor that reads air pressure continuously. Errors due to these readings should also be considered.
- Servo motor and mechanical structure errors – Inaccuracies in the structure used to track antennas must be considered and reduced as much as possible.
- Yaw and pitch correction – Whenever the UAV moves, it necessarily nods in the direction it is moving to, thus causing a misalignment in the UAV antenna.

Most of the noted errors may lead to antenna misalignment and cause overall degradation of the link quality. This effect is analyzed using simulation results in the following subsection. Weather may also affect the operation of our system. In particular, weather conditions such as rain may become a serious source of attenuation at frequencies of above 5 GHz. However, we can rely on existing measurements to predict this impact. For example, in [48] the authors found that rain constitutes 0.2 to 3.7 dB/km of attenuation for a similar frequency for rainfall rates ranging from 5 (light rain) to 132 (heavy rain) mm/h. Water vapor does not significantly contribute to attenuation at this frequency [49]. Watery snow conditions have slightly stronger attenuation ranging up to 15 dB/km [50].

Given our equipment, we may conclude that the effect of rain may be neglected in the desired operating scenario. Going further, we plan to investigate the impact of wind on the steering system, whenever the corresponding weather conditions will be met in a live deployment.

C. Simulating Misalignment Errors

To complement the above considerations, we prepared a dedicated simulation script to observe and characterize the impact of one of the main sources of errors in our algorithm: steering misalignment. Other types of errors, such as inaccuracies produced by the compass or altitude barometer that are inherent to these devices are difficult to characterize, since their relative effect lays within the absolute value of the measurement results. A scenario with a drone following a predefined trajectory was created to simulate the impact of misalignment errors.

First, 10 reference points were selected for the trajectory, where the drone followed a straight line from point $[500, 150]$ to point $[500, 150]$. The altitude was maintained at $10 \pm 0.5$ meters. The movement speed was set to $10$ m/s so that the simulation results closely correspond to our experimental scenario. In
the simulations, we assigned a misalignment error of $5^\circ$ in both planes. Hence, two randomly chosen sets of values distributed between $-2.5$ and $2.5$ were generated. An example for errors in the azimuth plane is shown in Fig. 6(a). This misalignment angle is expected to appear due to weather conditions or mobility problems. The resulting received power data is shown in Fig. 6(b). Note that the received power decreases because the drone moves farther away from the ground station during the simulations. It is clear that such a small angular fluctuation does not have a significant effect on propagation.

Furthermore, the antenna gains on both sides of the link were calculated from the radiation patterns for each antenna (described in Table 4) for the generated misalignment angles (see Fig. 6(c)). Note that the antenna gain is not the maximum gain of the antenna, but rather the value corresponding to the misalignment angle for the antenna pattern. This confirms that the practical impact of the ground station (more directivity and reduced effect due to wind) is much higher than that of the drone antenna due to the beamwidth of the main lobe. Hence, this confirms that the hardware selected for our experimental setup was appropriate.

**Table 4. Antenna parameters**

<table>
<thead>
<tr>
<th>Parameter</th>
<th>GSA</th>
<th>UAV antenna</th>
</tr>
</thead>
<tbody>
<tr>
<td>Manufacturer</td>
<td>MikroTik</td>
<td>MikroTik</td>
</tr>
<tr>
<td>Model</td>
<td>LGH 5</td>
<td>SXT Lite 5</td>
</tr>
<tr>
<td>Protocol</td>
<td>Nv2</td>
<td>Nv2</td>
</tr>
<tr>
<td>Transmit power</td>
<td>20 dBm</td>
<td>20 dBm</td>
</tr>
<tr>
<td>Maximum gain</td>
<td>24 dB</td>
<td>16 dB</td>
</tr>
<tr>
<td>Beamwidth</td>
<td>$7^\circ \times 7^\circ$</td>
<td>$25^\circ \times 25^\circ$</td>
</tr>
<tr>
<td>Frequency</td>
<td>5150 – 5875 MHz</td>
<td>4920 – 5920 MHz</td>
</tr>
<tr>
<td>Bandwidth</td>
<td>40 MHz</td>
<td>40 MHz</td>
</tr>
</tbody>
</table>

**D. Comparison with Existing Solutions**

In the following lines, the steering algorithm proposed in this study is compared with two other solutions to demonstrate and confirm its efficiency. A similar steering solution between two terrestrial robotic systems was proposed in [51]. The authors used a direction of arrival algorithm, which requires that an array of antennas continuously reads the phases of the received signals to compute the steering angle. In this way, a linear array of antennas can be used to obtain the steering angle in one direction (either azimuth or elevation), while a two-dimensional array can be used to obtain both the azimuth and the elevation steering angles simultaneously. In our system, we utilize our own algorithm to obtain both steering angles, which avoids the need for more than two antenna elements in the overall system.

Another solution for the tracking problem at hand is to use high resolution video cameras combined with an image processing algorithm to detect and track the target, as was considered in [46]. However, this method requires direct line-of-sight (LOS) to detect the target with high resolution images produced by the camera. Therefore, this solution is not preferred when considering scenarios where LOS can be blocked by surrounding objects. To the best of our knowledge, no other group has published any research on the automatic steering capabilities of the GSA and the aerial vehicle. We thus believe that our novel solution can be considered as a starting point for future implementations and improvements that target UAV-aided communication systems.

**IV. Numerical Results**

For a number of applications, high throughput becomes one of the primary metrics of interest [52–54]. To deliver high data rates to remote areas, it is necessary to use highly directional an-
tennas with relatively large transmit power. However, antenna beam misalignment will produce additional losses, which can affect the link quality. In this section, we study the impact of antenna misalignment depending on their radiation pattern. To analyze this appropriately, the radiation patterns from each antenna must be obtained. The GSA patterns were acquired from the manufacturer’s website\(^1\). However, the manufacturer did not provide patterns for the UAV antenna\(^2\), and it was necessary to approximate the radiation pattern from the antenna parameters. The maximum gain and beamwidth were taken into account, while a Chebyshev filter was used to approximate the final model. We assumed the elevation and azimuth patterns to be equal.

According to the parameters provided in Table 4, one can see that the drone’s main antenna lobe is wider than that of the GSA, which makes this antenna less sensitive to beam misalignment. From the radiation patterns, we can analytically obtain 3D plots for the misalignment losses, which also corroborate this claim. The theoretical plots are shown in Fig. 7 and 8, and these figures represent losses produced by misalignment in the vertical and horizontal planes, depending on the radiation pattern for each antenna. Higher losses will be produced due to misalignment for an antenna with higher directionality.

According to theory, the central part of the signal energy propagates in the first Fresnel zone \([55]\). This zone is an ellipsoid around the ideal direct ray. As an approximation, signal loss during propagation is considered to be free-space loss (FSL) if the first Fresnel zone is free from obstacles. However, propagation can also be assumed to follow FSL when there are obstacles within 60% of the radius of the first Fresnel zone.

We obtained the path loss (PL) while gathering our measurement capabilities of the selected hardware. The experiment was conducted in a suburban area under a LOS scenario over distances ranging from 0 to 550 m. The ground station antenna was fixed at one meter above ground and the drone was kept at approximately 3 m above ground. Measurements were gathered while mindful of the Fresnel zone to illustrate the propagation behavior of the signal.

---

1\(^{\text{See "LHG 5 datasheet" by MikroTik, 2018: https://i.mt.lv/routerboard/files/LHG-170927115805.eps}}\)

2\(^{\text{See "SXT 5 datasheet" by MikroTik, 2018: https://i.mt.lv/routerboard/files/sxt_5-150206095451.eps}}\)
ments, as shown in Fig. 10. The considered values were acquired at various distances and were compared to FSL [56]. During measurements, a simple script was executed at both sides of the link to read the received power.

Table 4 summarizes the values used within the power budget when both antennas were set to operate at 5500 MHz. The PL was calculated with the equation below using given values of $P_{tx}$, $P_{rx}$, $G_{tx}$, and $G_{rx}$ as

$$PL(dB) = P_{tx}(dBm) - P_{tx}(dBm) + G_{tx}(dB) + G_{rx}(dB).$$  \(7\)

We obtained a free-space model for our scenario based on the above results. As a first approach, measurements were compared with FSL, which can be approximated as recommended by 3GPP [57]. Since the measurements were performed in non-ideal scenarios, the obtained losses were considerably higher than those predicted from the FSL model. However, the PL follows the trend that would be observed in an ideal scenario.

$$L_{FSL}(dB) = 32.45 + 20\log_{10} \left( \frac{d(km)}{100} \right) + 20\log_{10} \left( f(MHz) \right).$$  \(8\)

Finally, we evaluated the link quality as a function of distance. A network connection was delimited by two hosts running iPerf v3, with the drone acting as a server and the ground station acting as a client. The evaluation was completed based on the following set of metrics (see Fig. 11):

- **Jitter** – variation in the delay between received packets. Measured based on iPerf UDP tests.
- **Packet losses** were also measured with iPerf UDP tests (<1%, if not, multiple retransmissions).
- **Throughput** was measured with iPerf via TCP tests under saturated conditions.

Overall, the measurements represent the mean and the 95% confidence interval for all plots (see Fig. 11). As one can see from the plots, the bandwidth delivered with our designed system is insensitive to communication distance in this scenario. In contrast, packet losses and jitter are more significantly affected by the environmental conditions, even though they still satisfy the target application requirements. Interestingly, while gathering measurements at around 120 meters, a truck was passing between the transmitter and the receiver. Its presence affected packet loss and jitter but had almost no impact on bandwidth.

To this end, Fig. 11(c) reveals that the link budget of a wireless connection allows throughput to be maintained with even further increased data losses due to an Ethernet limitation of 100 Mbit/s. Overall, we show that our developed framework may deliver high throughput and low latency communication in mission-critical and on-demand scenarios.
VI. Conclusions and Lessons Learned

In this paper, we developed a prototype of an aerial base station. This system has been discussed at length in the past, but was never thoroughly investigated in the field trials due to a number of challenges. We presented a working prototype of the UAV base station, which may carry a microwave antenna and a steering beam alignment mechanism that keeps the antenna pointed toward the ground station. The steering mechanism operation is the most challenging and crucial component in our solution. Since our measurements were collected for link lengths ranging up to 550 m, and the microwave connection used an IEEE 802.11a/n-like protocol, the resulting accuracy was sufficient as a starting point. In the future, we plan to employ millimeter-wave equipment, which will require a more precise steering system.

We also presented our initial measurements, which were gathered with the constructed prototype. The results were collected in an outdoor scenario and were further compared with the theoretical free-space model. The trend in the measured losses is similar to theoretical expectations, but losses shift to higher values due to realistic weather conditions. In a future study, we plan to utilize gas-powered UAVs in a subsequent phase of our development. This should allow us to test not only the backhaul link itself, but also the downlink from the UAV to potential ground stations. One could subsequently determine the equipment requirements and develop a full-scale proof-of-concept system.

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REFERENCES


Jiri Pokorný is a Ph.D. candidate in the Department of Telecommunications at Brno University of Technology, Czech Republic. He received his M.Sc. in electronics and communications (2016) from the same university. His research topics include communications in the mmWave spectrum, Unmanned Aerial Vehicles, and NB IoT technology.

Alekandr Ometov is a Ph.D. candidate at Tampere University of Technology’s Laboratory of Electronics and Communications Engineering. He holds an M.Sc. with distinction in telecommunications (2016) from the same university and a Specialist degree (2013) in information security from St. Petersburg State University of Aerospace Instrumentation. His research interests include wireless communications, information security, heterogeneous networking, cooperative communications, and machine-to-machine applications.

Pablo Pascual Campo received the Bachelor’s degree with distinction in Telecommunications and Electrical Engineering from the Department of Signals, Systems and Radiocommunications, Universidad Politecnica de Madrid (UPM), Spain, in 2016. He is currently studying for his Master’s degree specialization in RF systems and signal processing at Tampere University of Technology (TUT), Finland. His major research interests are wireless communications, RF circuits, MIMO, and full-duplex systems and applications.

Carlos Baquero Barneto received the B.Sc. degree in Telecommunication Engineering from Universidad Politécnica de Madrid, Madrid, Spain. He is currently an M.Sc. student in the Laboratory of Electronics and Communications Engineering at Tampere University of Technology (TUT), Tampere, Finland. His general research interests include RF circuits and unmanned aerial vehicle systems.
Pavel Masek received his M.S. and Ph.D. degrees in electrical engineering from the Faculty of Electrical Engineering and Communication, Brno University of Technology, Czech Republic, in 2013 and 2017, respectively. His work on a variety of networking-related topics has been published in internationally recognized venues, including IEEE Communications Magazine, and he has developed several technology products. His primary research interest lies in the area of M2M/H2H/D2D communication, next-generation cellular networks, heterogeneous networking, and data offloading techniques.

Alexander Pyattaev is a CTO of Yl-Verkot Oy, Finland. He received his Ph.D. from the Department of Electronics and Communications Engineering at Tampere University of Technology (TUT), Finland. He received his B.Sc. degree from St. Petersburg State University of Telecommunications, Russia, and his M.Sc. degree from TUT. Alexander’s work on a variety of networking-related topics has been published in internationally recognized venues, and he holds several technology patents. His primary research interest lies in the area of future wireless networks, particularly in shared spectrum access, smart RAT selection, and flexible and adaptive topologies.

Ainoa Garcia Marti received her B.S. in Telecommunication Systems Engineering from Universidad Politecnica de Madrid, Spain, in 2018. Her undergraduate thesis at Tampere University of Technology is related to low Earth satellite communication systems. She currently works with YL-Verkot Oy as a trainee.

Carlos Castillo Mateos received his B.S. degree in Telecommunication Engineering from Universidad Carlos III, Spain, in 2016. He is currently working toward an M.S. degree in Information technology, communication systems and Networking in Tampere University of Technology, Finland, and is working as a researcher with Yl-Verkot Oy.

Sergey Andreev is a senior research scientist in the Laboratory of Electronics and Communications Engineering at Tampere University of Technology, Finland. He received his Specialist degree (2006) and Cand.Sc. degree (2009) from St. Petersburg State University of Aerospace Instrumentation, Russia, as well as his Ph.D. degree (2012) from Tampere University of Technology. He has (co-)authored more than 150 published research papers on wireless communications, energy efficiency, heterogeneous networking, cooperative communications, and machine-to-

Jiri Hosek received his M.S. and Ph.D. degrees in electrical engineering from the Faculty of Electrical Engineering and Communication at Brno University of Technology (BUT), Czech Republic, in 2007 and 2011, respectively. He is currently working as an associate professor and senior researcher at the Department of Telecommunications, BUT. His research work has concentrated on industry-oriented R&D projects in the area of future mobile networks, the Internet of Things, and home automation services. He has (co-)authored more than 70 research works on networking technologies, wireless communications, quality of service, user experience, and IoT applications.

Yevgeni Koucheryavy is a full professor at the Laboratory of Electronics and Communications Engineering of Tampere University of Technology. He received his Ph.D. degree (2004) from the same university. He is the author of numerous publications in the field of advanced wired and wireless networking and communications. His current research interests include various aspects of heterogeneous wireless communication networks and systems, the Internet of Things and its standardization, and nanocommunications. He is an Associate Technical Editor of IEEE Communications Magazine and an editor of IEEE Communications Surveys & Tutorials.